Attitude & Heading Reference System AHRS-II-P

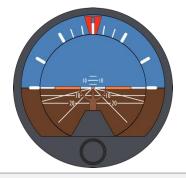


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The **Inertial Labs Attitude and Heading Reference Systems, AHRS-II** is the next generation of enhanced, highperformance strapdown systems that determines absolute orientation (**Heading, Pitch and Roll**) for any device on which it is mounted. Orientation is determined with high accuracy for both motionless and dynamic applications.



The Inertial Labs AHRS-II-P utilizes Tactical-grade IMU: 3-axes each of precision accelerometers & gyroscopes and Ultra High precision, gyro-compensated, embedded Fluxgate Compass to provide accurate Heading/Yaw, Pitch and Roll of the device under measure.

Integration of gyroscopes' output provides high frequency, real-time measurement of the device rotation about all three rotational axes. Accelerometers and Fluxgate magnetometer measure absolute Pitch, Roll and magnetic Azimuth at AHRS initial alignment as well as providing ongoing corrections to gyroscopes during operation.

KEY FEATURES AND FUNCTIONALITY

- State-of-the-art Kalman Filter based algorithms for different dynamic motions of UAV, Aircrafts, Helicopters, Vessels, Ships, ROV, UUV, Robots, UGV, AGV, Gimbals and Antennas
- Highly accurate, sensitive, and temperature stable Fluxgate magnetometers (in-house technology)
- Input from External GNSS Receiver;
- 0.6 deg Gyro-Stabilized Heading and 0.08 deg Pitch & Roll Dynamic Accuracy over operational temperature range
- 1 deg/hr gyroscopes Bias in-run stability
- 5 µg accelerometers Bias in-run stability
- Suitable for Primary Attitude Reference and as Motion Control System for marine applications
- Embedded 2D and 3D magnetic calibration on hard and soft iron
- RS-232, RS-422, and CAN 2.0 interfaces
- Full temperature calibration of all sensing elements
- Up to 200 Hz data update rate (AHRS) and 2000 Hz data rate (IMU)
- Compact design. Environmentally sealed (IP67)

One of the key elements to the success of Inertial Labs AHRS-II is its use of **Inertial Labs 8mm Fluxgate Magnetometers**, which have distinct advantages over commonly used magneto-inductive or magneto-resistive magnetometers. In operation over time and temperature fluxgate magnetometers have superior stability and repeatability. In terms of sensitivity, fluxgate magnetometers provide up to two orders of magnitude increased sensitivity.

In addition to the performance advantages, unlike the chip-level magnetometer technology, fluxgate magnetometer technology has been depended on for over 70 years to provide an accurate reference to North. It remains the most reliable magnetic sensor technology for determining an object's heading.

AHRS-II-P key specifications summary

Parameter	Units			
Status		In production (Since 2006)		
Heading static accuracy	deg	0.3		
Heading dynamic accuracy	deg	0.6 (gyro magnetic)		
Heading dynamic accuracy	deg	0.1 (with GNSS input)		
Pitch & Roll static accuracy	deg	0.05		
Pitch & Roll dynamic accuracy deg		0.08		
Gyroscopes Bias in-run stability	deg/hr	1		
Accelerometers Bias in-run stability	mg	0.005		

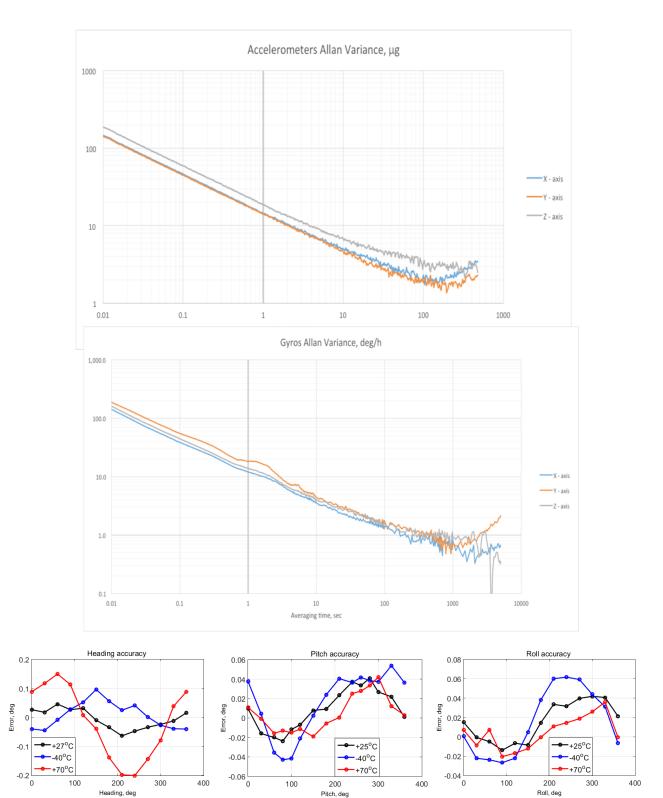
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Inertial Labs AHRS-II-P key performance

Parameter Status	Units	AHRS-II-P In production			
Status	Without GNSS	In production Euler angles; Quaternion; Relative Altitude; IMU data (angular rates,			
	input	accelerations); Magnetic field; Pressure; Delta Theta, Delta Velocity			
Output signals	With GNSS input	Horizontal & Vertical Positions (LAT, LONG); Heading, Pitch, Roll, Velocity			
	with Gives input	PPS time, IMU data (angular rates, accelerations)			
Input signals		External GNSS receiver, External magnetic compass			
Update rate (AHRS data)	Hz	1 200 (user settable)			
Update rate (IMU data)	Hz	2000			
Start-up time Full Accuracy Data (Warm-up Time)	sec	<1 10			
Heading	Units	10			
Range	deg	0 to 360			
Angular Resolution	deg	0.01			
Static Accuracy Over Operational Temperature Range	deg RMS	0.3			
Dynamic Accuracy Over Operational Temperature Range	deg RMS	0.6			
Dynamic Accuracy with GNSS input	deg RMS	0.1			
Pitch and Roll	Units				
Range: Pitch, Roll	deg	±90, ±180			
Angular Resolution	deg	0.01			
Static Accuracy in whole Temperature Range Dynamic Accuracy Over Operational Temperature Range	deg deg RMS	0.05			
Dynamic Accuracy Over Operational Temperature Range Dynamic Accuracy with GNSS input	deg RMS	0.08			
Relative altitude	Units	0.05			
Measurement range	meters	-600 to 9000			
Resolution	meters	0.01			
Relative accuracy	meters	<1			
Gyroscopes	Units				
Туре		Tactical-grade			
Measurement range	deg/sec	±450 / ±950 / ±2000			
Bandwidth Bias in-run stability (RMS, Allan Variance)	Hz deg/hr	262			
Bias residual error over operational temperature range	deg/hr, RMS	<35			
SF accuracy	ppm	1000			
Noise (Angular Random Walk)	deg/√hr	<0.2			
Non-linearity	ppm	100			
Axis misalignment	mrad	0.15			
Accelerometers	Units				
Туре		Tactical-grade			
Measurement range	g	±8 g / ±15 g / ±40 g			
Bandwidth	Hz	262			
Bias in-run stability (RMS, Allan Variance)	mg	$\frac{0.005 (\pm 8 \text{ g}) / 0.02 (\pm 15 \text{ g}) / 0.03 (\pm 40 \text{ g})}{0.5 (\pm 8 \text{ g}) / 0.7 (\pm 15 \text{ g}) / 1.2 (\pm 40 \text{ g})}$			
Bias error over temperature range (RMS) Bias one-year repeatability	mg	0.5 (±8 g) / 0.7 (±15 g) / 1.2 (±40 g) 1.0 (±8 g) / 1.3 (±15 g) / 1.5 (±40 g)			
Velocity Random Walk	mg m/s/√hr	0.015 (±8 g) / 0.035 (±15 g) / 0.045 (±40 g)			
SF accuracy	ppm	150 (±8 g) / 300 (±15 g) / 500 (±40 g)			
Non-linearity	ppm	100			
Axis misalignment	mrad	0.15			
Magnetometers (Embedded Fluxgate Compass)	Units				
Measurement range	Gauss	±1.6			
Bias in-run stability, RMS	nT	0.2			
Noise density, PSD	nT√Hz	0.3			
SF accuracy	%	0.02			
Bandwidth	Hz Units	50			
Pressure Measurement range	hPa	300 - 1100			
Bandwidth	Hz	50			
Resolution	Pa	2			
Bias in-run stability (RMS, Allan Variance)	Pa	0.8			
Noise density	Pa/√Hz	1			
Environment	Units				
Operating temperature	deg C	-40 to +70			
Storage temperature	deg C	-50 to +85			
MTBF (G _M @ +65degC)	hours	100,000			
Vibration & Shock		MIL-STD-810G MIL-STD-461			
Electrical EMC/EMI	Units				
Supply voltage	V DC	9 to 36			
Power consumption	Watts	1.4 (2.4 with data logger)			
Output Interface	-	RS-232, RS-422, CAN 2.0			
Output data format	-	Binary, NMEA 0183 ASCII characters			
Physical	Units				
Size	mm	120 x 50 x 53			

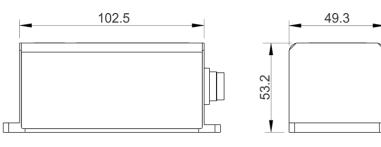
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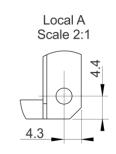


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AHRS-II-P mechanical interface drawing



120.5 ø 4.4 \oplus φ Local A



Notes:

1. All dimensions are in millimeters. All dimensions within this drawing 2. are subject to change without notice. Customers should obtain final drawings before designing any interface hardware.

Connector type: Binder Series 723. Male receptacle, 24 pin, shielded, rearmounting

Model	Gyro	Accel	Calibration	Connector	Color	Data Logger	Version	Interface
AHRS-II-P	G450	A8	TMGA	C3 (default)	B (default)	S64 (option)	V1	1
	G950	A15		. ,	D			2
	G2000	A40			G			4
			W			5		
								11
								22
xample: AHR	S-II-P-G450-A8	-TMGA-C3B-V1	.11					145

Description:

- AHRS-II-P: Professional Version of Attitude and Heading Reference System
- G450: Gyroscopes measurment range = \pm 450 deg/sec G950: Gyroscopes measurment range = \pm 950 deg/sec
- G2000: Gyroscopes measurement range = ±2000 deg/sec
- A8: Accelerometers measurement range = ± 8 g
- A15: Accelerometers measurement range = ± 15 g
- Ado: Accelerometers measurement range = ±40 g TMGA: Magnetometers, Gyroscopes and Accelerometers C3: 24 pins connector (RS-232, RS-422, CAN, Ethernet interfaces)
- B Black Color (default)
- D Desert Color (Desert tan, color code 33446 (tan 686A) per FED-STD-595, Change Notice 1.)
- G Green W White
- S64: 64GB embedded Data Logger (optional)
- VX.1: RS-232 interface VX.2: RS-422 interface
- VX.4: CAN interface
- VX.5: Ethernet interface
- VX.11: two RS-232 interfaces
- VX.22: two RS-422 interfaces VX.145: RS-232, CAN and Ethernet interfaces
- VX.245: RS-422, CAN and Ethernet interfaces

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